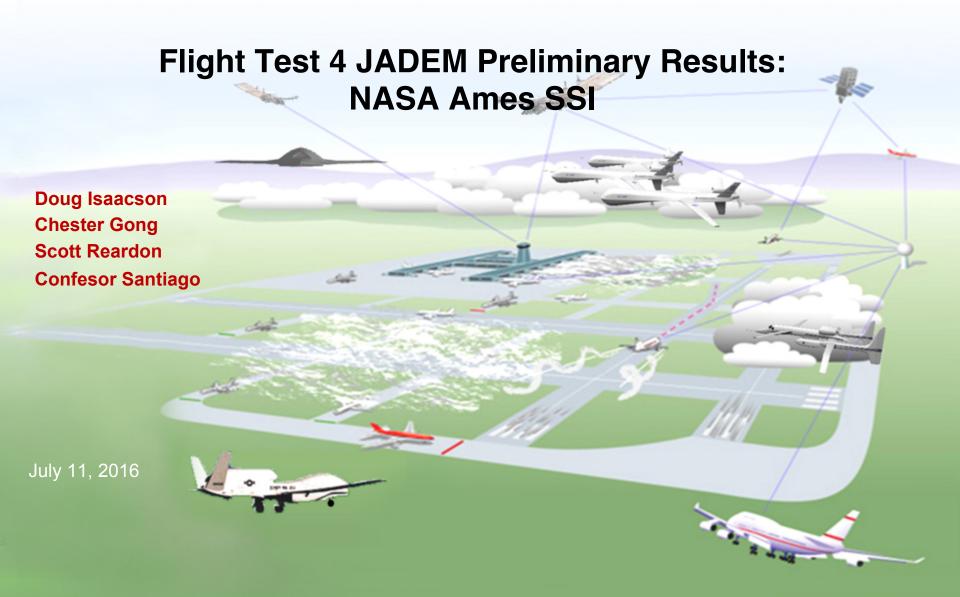


National Aeronautics and Space Administration



Outline



- System Under Test: JADEM Alerting and Guidance
- NASA Ames SSI Objectives and Scenario Preparation
- Sample Scenario Analysis
- Preliminary Findings
 - FT4 Evaluation Criteria Compliance
 - DAA Alert Timing Analysis
 - Unexpected TCAS RAs
 - Mode C Intruder Guidance Stability
 - Well Clear Recovery Performance
- Lessons Learned & Recommendations



JADEM Alerting and Guidance

Flight Test 4 – Alerting Criteria



Symbol	Name	Pilot Action	Buffered Well Clear Criteria	Alerting Time Threshold	Aural Alert Verbiage
A	TCAS RA	• Immediate action required • Comply with RA sense and vertical rate • Notify ATC as soon as practicable after taking action (Drive		х	"Climb/ Descend"
	DAA Warning Alert	 Immediate action required Notify ATC as soon as practicable after taking action 	DMOD = 0.75 nmi HMD = 0.75 nmi ZTHR = 450 ft modTau = 35 sec	25 sec (TCPA approximate: 60 sec)	"Traffic, Maneuver Now"
	Corrective DAA Alert	 On current course, corrective action required Coordinate with ATC to determine an appropriate maneuver 	DMOD = 0.75 nmi HMD = 0.75 nmi ZTHR = 450 ft modTau = 35 sec	55 sec (TCPA approximate: 90 sec)	"Traffic, Avoid"
	Preventive DAA Alert On current course, corrective action should not be required Monitor for intruder course changes Talk with ATC if desired		DMOD = 1.0 nmi HMD = 1.0 nmi ZTHR = 700 ft modTau = 35 sec	55 sec (TCPA approximate: 90 sec)	"Traffic, Monitor"
A	Remaining Traffic	No action expected	Within surveillance field of regard	Х	N/A

Cooperative/Non-cooperative Alerting Structure



Cooperative Aircraft						
Symbol	Symbol Name					
A	TCAS RA					
	DAA Warning Alert	"Traffic, Maneuver Now"				
	Corrective DAA Alert	"Traffic, Avoid"				
	Preventive DAA Alert					
A	None (Target)	N/A				

Non-Cooperative Aircraft						
Symbol	Name	Aural Alert Verbiage				
	DAA Warning Alert	"Traffic, Maneuver Now"				
	Corrective DAA Alert	"Traffic, Avoid"				
	Preventive DAA Alert	"Traffic, Monitor"				
A	None (Target)	N/A				

Alerting Hysteresis



- <u>Section 2.2.4.2.4 Alerting</u>: Any DAA alert **shall** persist for a minimum of four seconds, unless the intruder is declared a higher priority alert
- <u>Evaluated</u> = alert level evaluated by the algorithm based on state at specific time epoch
- <u>Declared</u> = alert level declared to pilot based on hysteresis requirement

Corrective alert brotonget	ective alert prolon	ged
----------------------------	---------------------	-----

Warning alert prolonged

Preventive alert prolonged

Evaluated	Declared
preventive	preventive
corrective	corrective
corrective	corrective
corrective	corrective
preventive	corrective
preventive	corrective
preventive	preventive
warning	warning
preventive	warning
preventive	warning
warning	warning
preventive	preventive
preventive	preventive
clear	clear
clear	clear

Time

1

3

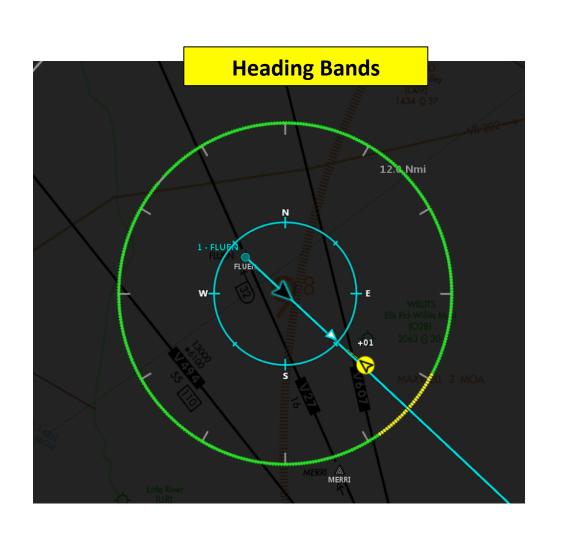
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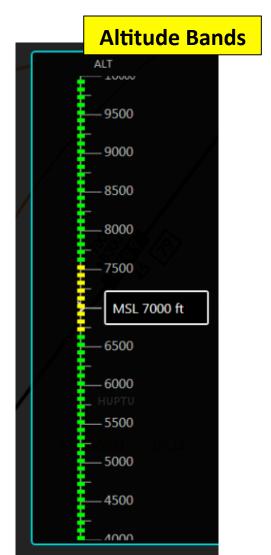
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OmniBands





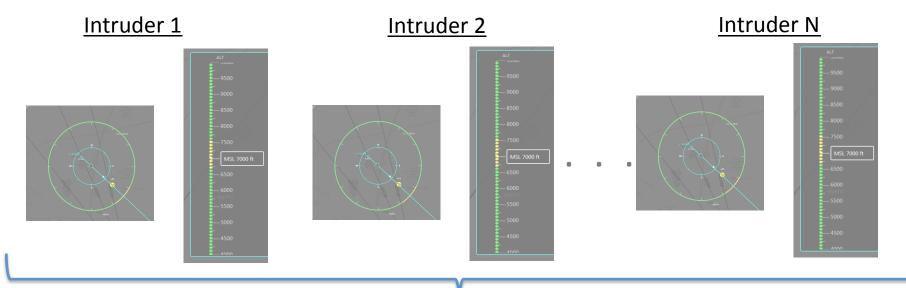
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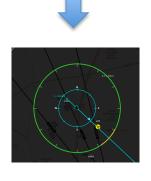
Guidance for Multiple Intruders



 Handles guidance for every intruder within field of view; generating individual heading and altitude bands based on alerting criteria for each intruder, then displaying the union of all bands into a single heading and altitude band



Integrated together for all Intruders







To display



Other Features (Constraints)

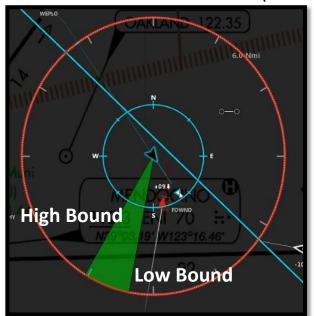


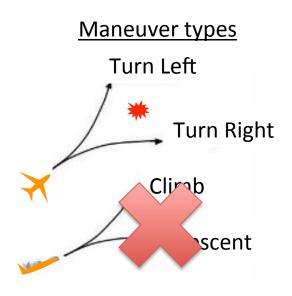
- In calculating heading bands, considers a single user-specified turn rate for ownship aircraft (e.g. 3 degrees per second)
- In calculating altitude bands, considers a single user-specified climb and decent rate for ownship aircraft (e.g. 1,000 FPM)
- Limits heading bands by:
 - User-specified discretization: e.g. 1-degree intervals
 - User-specified heading range: e.g. +/- 110 degrees
- Limits altitude bands by:
 - User-specified discretization: e.g. 100 ft intervals
 - User-specified relative altitude range: e.g. +/- 3,000 ft
- Band Colors
 - Green Banding = heading/altitude is not predicted to lead to a loss of well clear
 - Yellow Banding = heading/altitude is predicted to cause a DAA Corrective Alert (i.e. potential loss of well clear in the next 25-55 seconds)
 - Red Banding = heading/altitude is predicted to lead to a DAA Warning Alert (i.e. potential loss of well clear in 25 seconds or less)

Well Clear Recovery Algorithm



- Well-Clear Recovery (WCR) provides guidance to regain Well Clear when no maneuvers are found to remain well clear (i.e., otherwise all-red OmniBands)
- WCR guidance is based on a general purpose conflict detection and resolution algorithm (GRACE, modified for FT4)
 - Evaluates multiple intruders for conflicts (threats) based on user-defined separation standards
 - Supports guidance provided by JADEM (e.g., for PT4/PT5/PT6, IHITL, OmniBands, WCR)
 - Generates a conflict avoidance maneuver of left, right (no vertical for FT4)
 - Selects the lowest cost conflict-free solution if one is found
 - Selects the lowest cost (conflicting) solution otherwise





Method: Well Clear Recovery Cost Function



- Near mid-air collision (NMAC) cost
 - Penalizes all maneuvers too close to NMAC violation for any intruder, i.e. maximizes normalized separation at new, predicted closest point of approach
 - This cost naturally dominates when close to collision, which is the case in WCR
- Maneuver change cost
 - Penalizes frequent changes of maneuver types
 - Can improve guidance stability in the presence of noise
- Maneuver type (rank) cost
 - Favors "right-of-way-like" compliant maneuvers
- Specific maneuver type costs (preferences)
 - Can selectively enforce or suppress specific maneuver types (e.g. NO CLIMBS)
- Maneuver strength cost
 - Penalizes too aggressive maneuvers (magnitude of heading or vertical speed change)
- Maneuver duration cost
 - Penalizes long deviations from flight plan

DAA/TCAS RA interoperability



- Any intruder with an active corrective RA should be removed from all DAA guidance calculations
 - Horizontal DAA guidance will be shown for non-RA aircraft
 - All DAA vertical guidance should be suppressed during a corrective RA to prevent showing conflicting guidance to the pilot
- During a preventive RA, TCAS guidance should be an input to the DAA vertical guidance so that it is consistent (e.g. DO NOT CLIMB)
- Well clear recovery is limited to horizontal only
 - Prevents pilots from making maneuvers near the collision avoidance boundary which may degrade TCAS II performance
 - For both 'cooperative' and 'non-cooperative' intruders
- Any time ownship's compliance with a corrective RA leads to a secondary DAA Warning alert (maneuver now), DAA guidance shall revert to well clear recovery in order to be more direct with guidance, e.g.:
 - Compliance with TCAS 'DESCEND' RA leads to a secondary DAA Warning
 - Rather than show pilot full OmniBands suggestive guidance, limited suggestive guidance is displayed (e.g. maneuver left)



NASA Ames SSI FT4 Test Objectives and Scenario Preparation

NASA Ames SSI FT4 Objectives



- 1. Validate DAA requirements in stressing cases that drive MOPS requirements, including: High-speed cooperative intruder, Low-speed non-cooperative intruder, high vertical closure rate encounter, and Mode C/S-only intruder (i.e. without ADS-B).
- 2. Validate TCAS/DAA alerting and guidance interoperability concept in the presence of realistic sensor, tracking and navigational errors and in multiple-intruder encounters against both cooperative and non-cooperative intruders.
- 3. Validate 'Well Clear Recovery' guidance in the presence of realistic sensor, tracking and navigational errors.
- 4. Validate DAA alerting and guidance requirements in the presence of realistic sensor, tracking and navigational errors.
- 5. Collect data to support development and validation of trajectories specified in the DAA MOPS for DAA system acceptance testing.

Sample Scenario Template: 041-Am-S2M



FT4 TEST OBJECTIVES - SSIWest.2.g (FT4 encounter: 041)

Objective 2: Validate TCAS/DAA alerting and guidance interoperability concept in the presence of realistic sensor, tracking and navigational errors: (g) level-level TCAS RA encounters (mitigated, <10kft).

MOPS Sections 2.2.4.2.4 Alerting 2.2.4.3 Determine Processing, 2.2.4.4 **Collision Avoidance Interoperability**

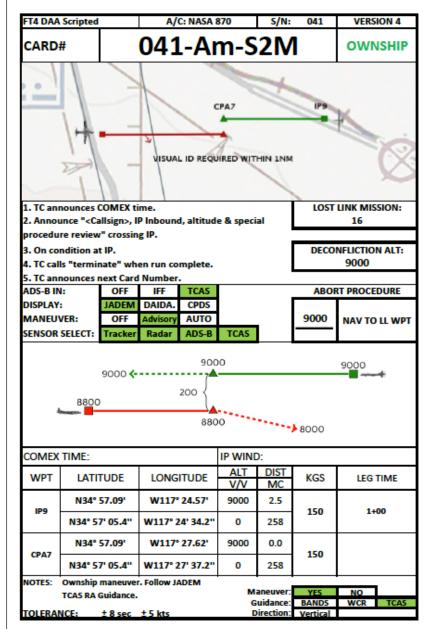
Configuration System Under Test: **Omnibands** Display: VSCS Contributing Sensors: 200 ft **TCAS** ADS-B Radar Tracker YES (S) YES YES YES Lateral Offset = 0.5 NM TCAS Alerting Boundary = 0.55 NM Minimum Altitude Offset: 200 ft **Ownship Test Objectives (TO)** 1. Validate DAA alert timing allows pilot sufficient time to assess options and maneuver to remain well clear. 2. Validate DAA guidance to UAS pilot is appropriate (reasonable, timely & responsive). Validate DAA alerting and guidance do not interfere with UAS pilot's ability to assess and follow TCAS RA guidance. Success Criteria 1. A timely (corrective initially) DAA alert is provided to the UAS pilot. A corrective TCAS RA is generated for the primary intruder and the UAS pilot complies with TCAS RA guidance. (retest if criteria not Data collected: LVC log file. SAAProc log files, TCAS log files for ownship and intruder, SAAP file ('raw' surveillance data). met) 3. **Test Method** MANUEVER: Pilot disregards DAA guidance and follows TCAS guidance if consistent with test constraints. Aircraft speeds (non-accelerating): Ownship 150 KGS, Intruder 180 KGS Encounter Length: 1 min (IP to CPA) Stable Conditions: stable at the IP (1 minutes prior to CPA) Test Termination Criteria: targets diverging, range > 0.75 nmi., no DAA alerts displayed. Climb/Roll/Pitch Rates: Ownship 0/0/0, Intruder -2500 FPM/0/0 Tolerance: ± 8 sec, ± 5 kts at IP crossing. **Evaluation Criteria**

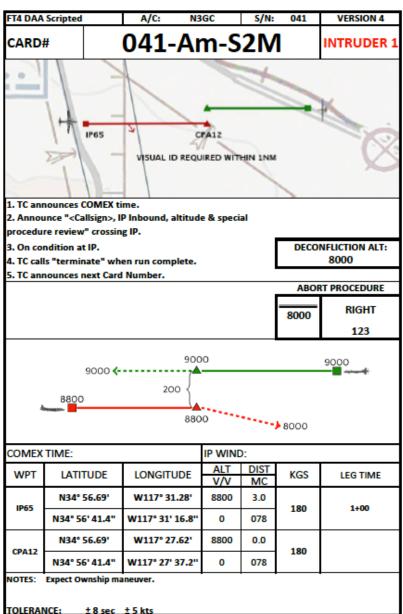
(Post-test analysis to determine if test objectives are met)

- UAS pilot receives DAA corrective alert with associated guidance.
 - Vertical DAA guidance indicates no viable vertical maneuvers in temporal proximity to TCAS RA.
 - DAA alerting and guidance for the primary intruder are removed while a TCAS RA is present for the primary intruder.
- DAA alerting and guidance is generated once RA expires (if appropriate). 4.
- DAA alert(s) and guidance are removed once ownship is clear of threat.

Sample Scenario Card: 041-Am-S2M







NASA Ames SSI FT4 Objectives



- Validate DAA requirements in stressing cases that drive MOPS requirements, including: High-speed cooperative intruder, Low-speed non-cooperative intruder, high vertical closure rate encounter, and Mode C/S-only intruder (i.e. without ADS-B). (18 encounters)
- Validate TCAS/DAA alerting and guidance interoperability concept in the presence of realistic sensor, tracking and navigational errors and in multiple-intruder encounters against both cooperative and non-cooperative intruders. (29 encounters)
- 3. Validate 'Well Clear Recovery' guidance in the presence of realistic sensor, tracking and navigational errors. (9 encounters)
- 4. Validate DAA alerting and guidance requirements in the presence of realistic sensor, tracking and navigational errors. (16 encounters)
- 5. Collect data to support development and validation of trajectories specified in the DAA MOPS for DAA system acceptance testing. (72 total encounters)



Sample Scenario Analysis 041-Am-S2M



Test Objective Assessment: 041-Am-S2M



FT4 TEST OBJECTIVES – SSIWest.2.g (FT4 encounter: 041)

Objective 2: Validate TCAS/DAA alerting and guidance interoperability concept in the presence of realistic sensor, tracking and navigational errors: (g) level-level TCAS RA encounters (mitigated, <10kft).

MOPS Sections 2.2.4.2.4 Alerting 2.2.4.3 Determine Processing, 2.2.4.4 Collision Avoidance Interoperability

Configuration System Under Test: **Omnibands** Display: VSCS Contributing Sensors: 200 ft **TCAS** ADS-B Radar Tracker YES (S) YES YES YES Lateral Offset = 0.5 NM TCAS Alerting Boundary = 0.55 NM Minimum Altitude Offset: 200 ft **Ownship Test Objectives (TO)** Validate DAA alert timing allows pilot sufficient time to assess options and maneuver to remain well clear. 2. Validate DAA guidance to UAS pilot is appropriate (reasonable, timely & responsive). Validate DAA alerting and guidance do not interfere with UAS pilot's ability to assess and follow TCAS RA guidance. Success Criteria 1. A timely (corrective initially) DAA alert is provided to the UAS pilot. A corrective TCAS RA is generated for the primary intruder and the UAS pilot complies with TCAS RA guidance. (retest if criteria not Data collected: LVC log file. SAAProc log files, TCAS log files for ownship and intruder, SAAP file ('raw' surveillance data). met) 3. **Test Method** MANUEVER: Pilot disregards DAA guidance and follows TCAS guidance if consistent with test constraints. Aircraft speeds (non-accelerating): Ownship 150 KGS, Intruder 180 KGS Encounter Length: 1 min (IP to CPA) Stable Conditions: stable at the IP (1 minutes prior to CPA) Test Termination Criteria: targets diverging, range > 0.75 nmi., no DAA alerts displayed. Climb/Roll/Pitch Rates: Ownship 0/0/0, Intruder -2500 FPM/0/0 Tolerance: ± 8 sec, ± 5 kts at IP crossing.

Evaluation Criteria (Post-test analysis to determine if test objectives are met)

- 1. UAS pilot receives DAA corrective alert with associated guidance.
- 2. Vertical DAA guidance indicates no viable vertical maneuvers in temporal proximity to TCAS RA.
- . DAA alerting and guidance for the primary intruder are removed while a TCAS RA is present for the primary intruder.
- 4. DAA alerting and guidance is generated once RA expires (if appropriate).
- . DAA alert(s) and guidance are removed once ownship is clear of threat.

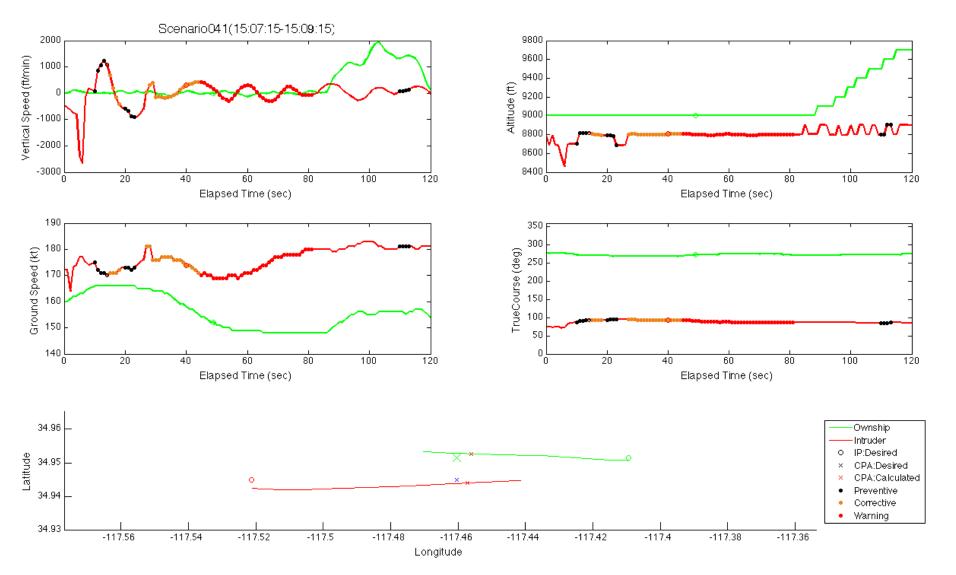
Scenario 041-Am-S2M: Event Data



Cycle Count	Intruder	Time	Clock Time	Event
191	N3GC	1461942437.39	15:07:17	FIRST_TRACK
198	N3GC	1461942445.37	15:07:25	DAA_PREVENTATIVE
203	N3GC	1461942450.38	15:07:30	DAA_CORRECTIVE
208	N3GC	1461942455.41	15:07:35	DAA_PREVENTATIVE
215	N3GC	1461942462.35	15:07:42	DAA_CORRECTIVE
233	N3GC	1461942480.40	15:08:00	DAA_WARNING
257	n/a	1461942504.39	15:08:24	WCR_TURN_RIGHT
258	N3GC	1461942505.38	15:08:25	LOWC
270	N3GC	1461942518.39	15:08:38	TCAS_CLIMB
291	N3GC	1461942539.39	15:08:59	СРА
291	N3GC	1461942539.39	15:08:59	TCAS_LEVEL_OFF
296	N3GC	1461942544.36	15:09:04	TCAS_AA_CLEAR
297	N3GC	1461942545.39	15:09:05	DAA_PREVENTATIVE
297	N3GC	1461942545.39	15:09:05	REGAIN_WC
300	N3GC	1461942548.40	15:09:08	FINAL_ALERT
308	N3GC	1461942556.36	15:09:16	LAST_TRACK

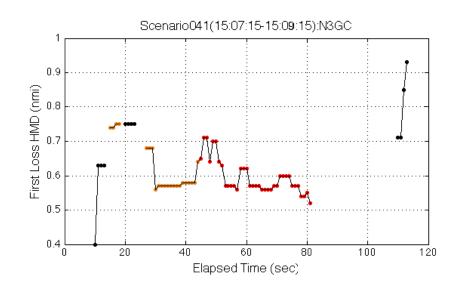
Scenario 041-Am-S2M: State Data

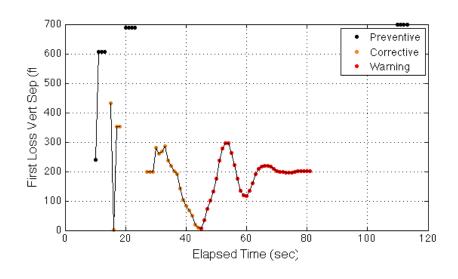


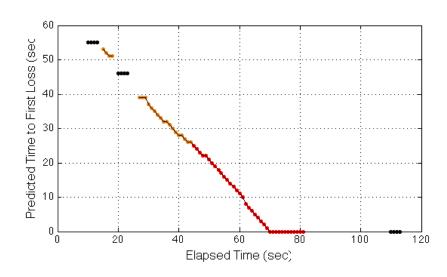


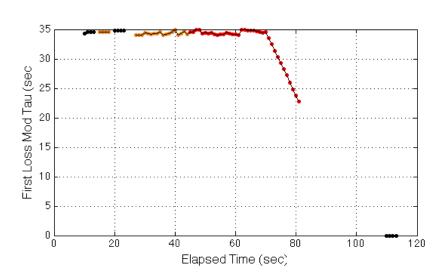
Scenario 041-Am-S2M: Threat Data





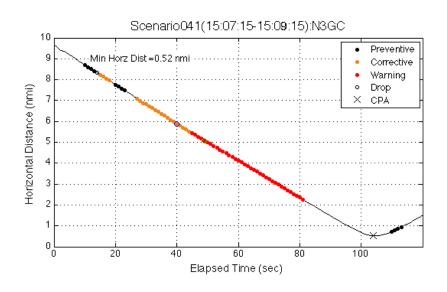


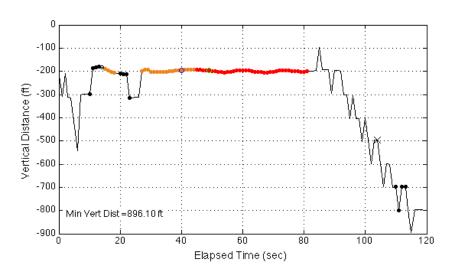


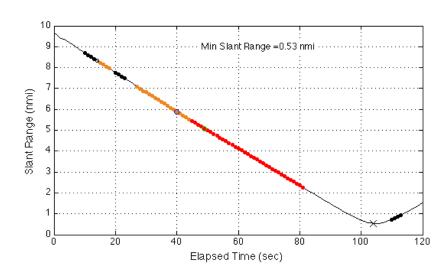


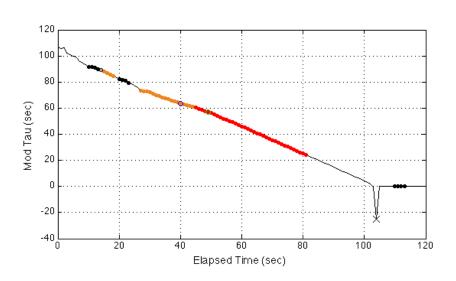
Scenario 041-Am-S2M: Encounter Data





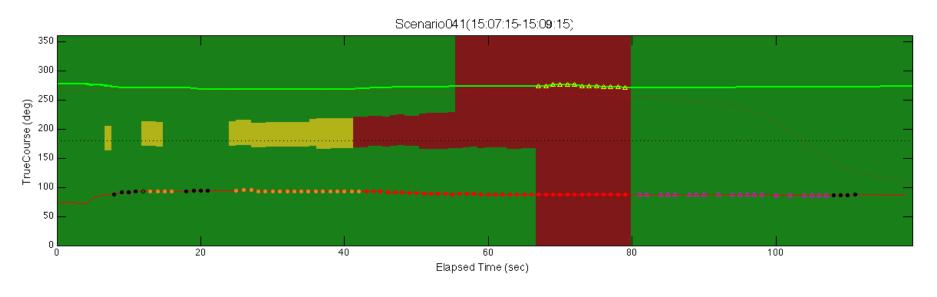


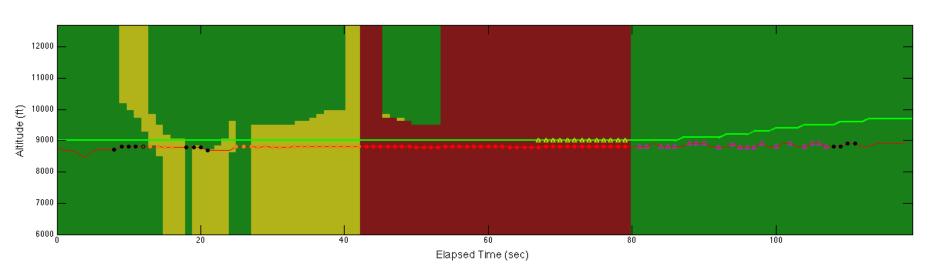




Scenario 041-Am-S2M: Guidance Data

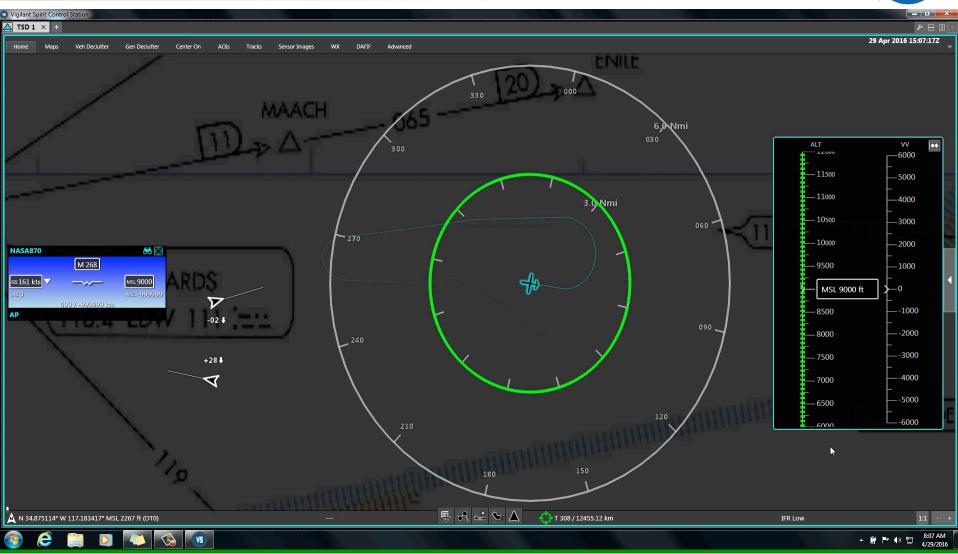






Scenario 041-Am-S2M: VSCS Recording







Preliminary Results

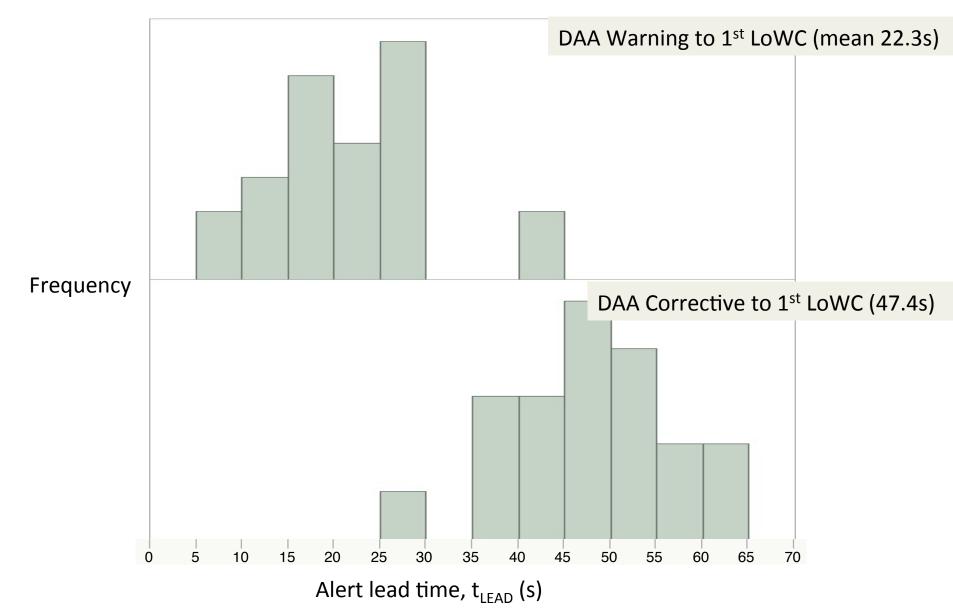
FT4 Evaluation Criteria Compliance



							WCR	
	Generated				No DAA Alert		maneuver and	
Scenario	Corrective	Remained Well	DAA Alerts	DAA Alerts	or Guidance	Generated WCR	Regain Well	Timely TCAS
number	Alert	Clear	Removed	After RA	During TCAS RA	Guidance	Clear	, Maneuver
7	TRUE	TRUE	TRUE	n/a	n/a	n/a	n/a	n/a
9	TRUE	TRUE	TRUE	n/a	n/a	n/a	n/a	n/a
10	TRUE	n/a	TRUE	n/a	n/a	n/a	n/a	n/a
12	TRUE	n/a	TRUE	n/a	n/a	n/a	n/a	n/a
13	TRUE	TRUE	TRUE	n/a	n/a	n/a	n/a	n/a
14	TRUE	FAISE	TRUE	n/a	n/a	n/a	n/a	n/a
15 16	TRUE TRUE	TRUE	TRUE TRUE	n/a n/a	n/a n/a	n/a n/a	n/a n/a	n/a n/a
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24	TRUE	n/a	TRUE	n/a	FALSE	n/a	n/a	TRUE
25	TRUE	n/a	TRUE	n/a	TRUE	n/a	n/a	TRUE
26 27	TRUE TRUE	n/a	TRUE TRUE	n/a	TRUE TRUE	n/a	n/a	TRUE TRUE
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43	TRUE	n/a	TRUE	n/a	FALSE	n/a	n/a	TRUE
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45 46	TRUE	n/a	TRUE	TRUE	FALSE	n/a	n/a	n/a
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63 64	TRUE TRUE	TRUE FALSE	TRUE TRUE	n/a	n/a	n/a	n/a n/a	n/a
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70	TRUE	n/a	TRUE	n/a	n/a	n/a	n/a	n/a
70 71	TRUE TRUE	n/a n/a	TRUE TRUE	n/a n/a	n/a n/a	n/a n/a	n/a n/a	n/a n/a
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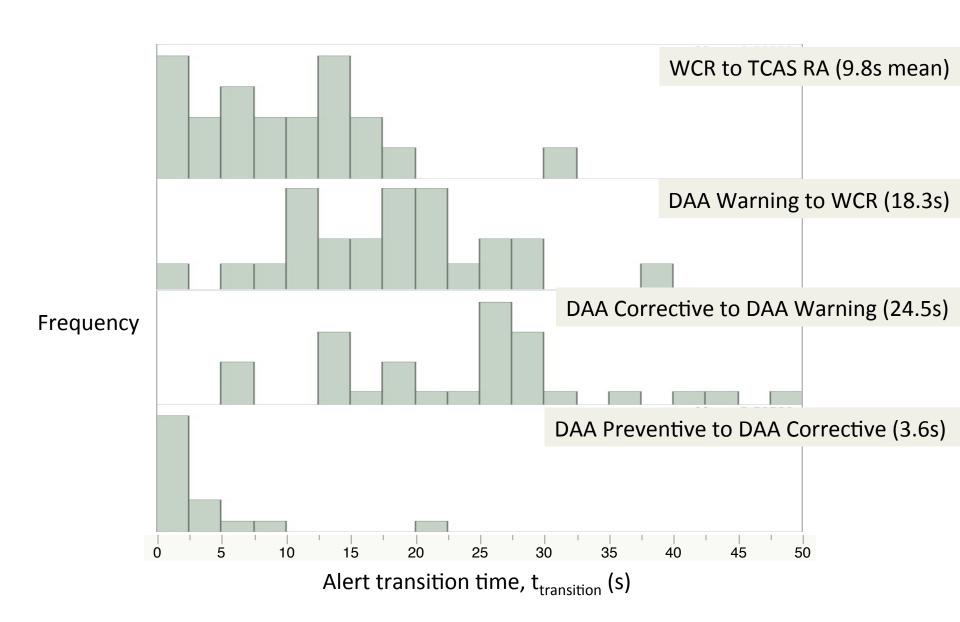
DAA Alert Lead Time (unmitigated)





DAA Alert Transition Time (unmitigated)





Unexpected TCAS RAs



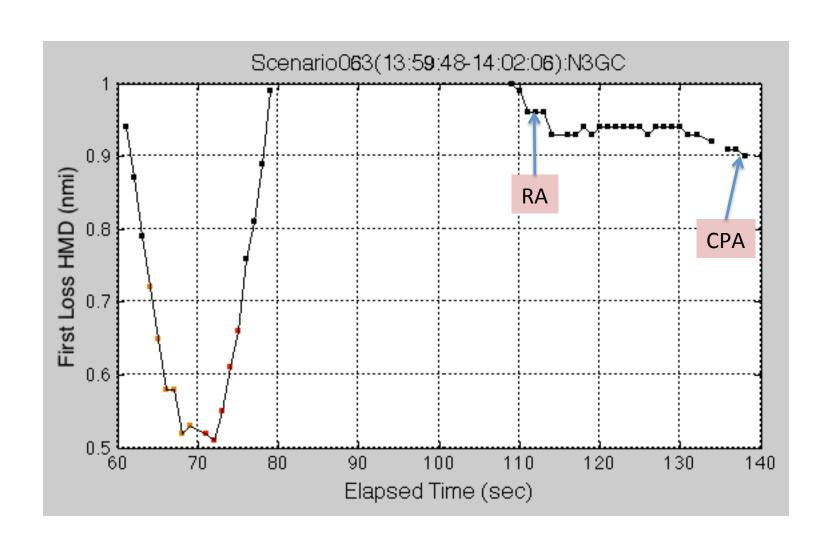
 Definition: a corrective RA is generated while DAA guidance and alerting indicates no maneuver is necessary to remain Well Clear

- Two observed unexpected TCAS RAs
 - Scenario 063 (maneuvering intruder to 45° intercept)
 - Scenario 047 (multi-intruder TCAS/WCR, primary intruder)
 - Both cases predicted HMD > 0.9nm at time of TCAS RA
 - Presumed TCAS bearing error led to TCAS RA while not meeting corrective or warning DAA HMD criteria (to be verified)

Sample Unexpected TCAS RA



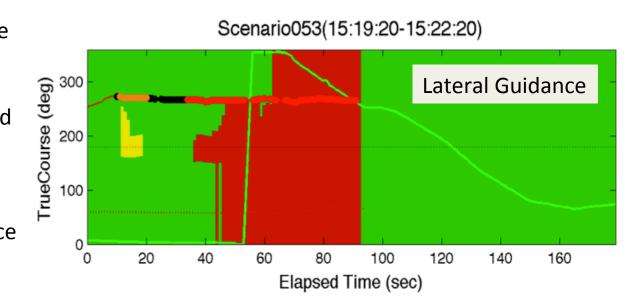
Scenario 063 (TCAS RA @ 112s elapsed time)

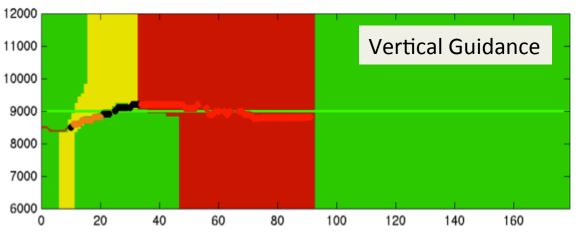


Guidance Stability for Mode C Intruders



- Some guidance instability was expected due to Mode C bearing error.
- Only 3 encounters included Mode C w/o ADS-B surveillance.
- Stable OmniBands guidance was observed in all 3 encounters.
- Bearing errors in integrated track are reduced within radar range.
- Intruder was within radar range for duration of encounter





Well Clear Recovery Performance



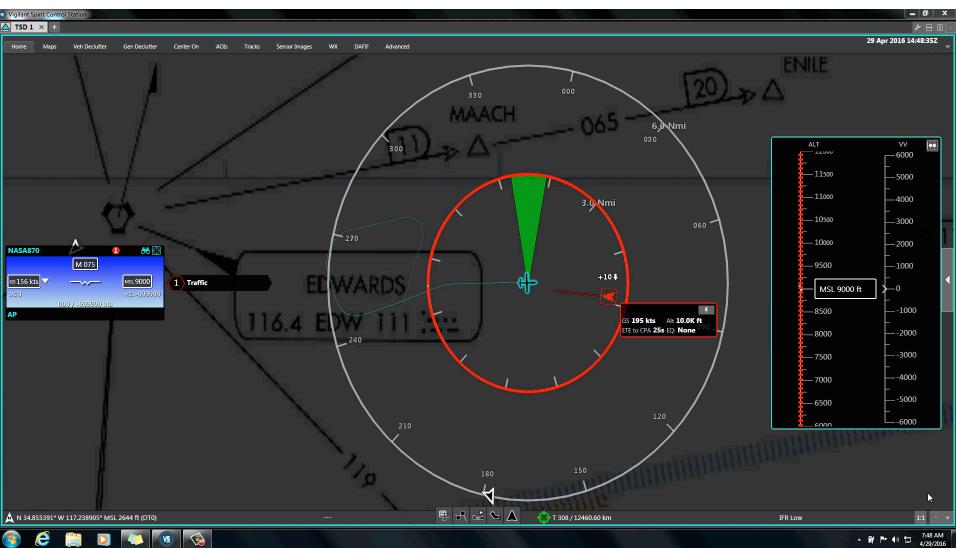
Well Clear Recovery (WCR) guidance was observed to be of limited utility



- Frequent changes to turn direction (left vs. right) of WCR guidance were observed for non ADS-B equipped intruders
- Short duration (<10s) between WCR onset and TCAS RA was observed in >50% of cases
- Large-turn WCR guidance aft of ownship was observed in some encounters.

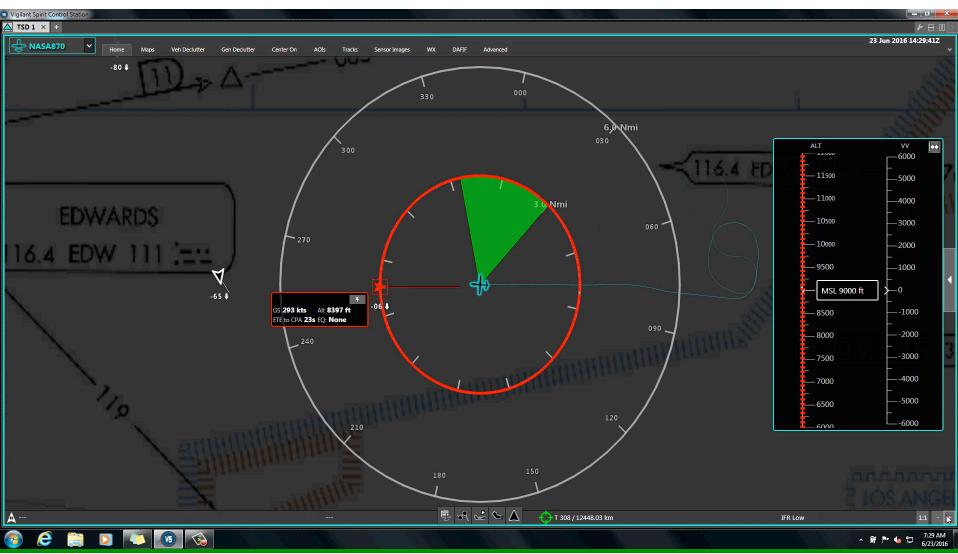
Sample WCR Instability: Short Duration





Sample WCR Instability: Direction Swap





Lessons Learned and Recommendations



- A wide array of encounters were successfully executed to evaluate the performance of prototype DAA alerting and guidance logic in a realistic environment... big thanks to entire FT4 team. Test criteria were largely met.
- Alert timing was largely acceptable, providing ample time for pilot to evaluate guidance and maneuver aircraft in most encounters -> acceptable alerting thresholds

Lessons Learned and Recommendations (cont.)



- WCR guidance was of limited utility for intruders lacking ADS-B.
 - WCR guidance should select a direction and maintain guidance except in cases of overwhelming evidence to the contrary
 - WCR guidance should include reasonable turn limits (e.g., no >90° turns)
 - Lead time from WCR to TCAS RA observed <10s in more than half of cases
 - WCR performance improved with ADS-B surveillance
 - May need to revisit TCAS interoperability concept regarding differences between
 ADS-B equipped and non-equipped intruders for WCR
- Stability of guidance for Mode C intruders appears adequate, but further investigation is warranted due to the limited sample size, particularly for highspeed, Mode C intruders.
- Unexpected TCAS RAs are possible due to differences in predicted HMD for DAA and TCAS... mitigations to prevent unexpected RAs may be required.

Next Steps



- Analyze trajectory prediction performance for each intruder sensor combination (i.e. Horizontal & Vertical CPA errors vs. time-to-CPA)
- Validate Mode C (non ADS-B) intruder guidance stability
- Develop and assess enhanced WCR guidance logic
 - Enhanced directional guidance stability
 - Turn angle limitation
 - Different logic for intruders with/without ADS-B
- Analyze scenarios from last 2 flights (17 scenarios)
- Investigate 'TCAS RA dropouts' to determine if new WCR guidance requirement(s) are necessary
- Further investigate 'unexpected TCAS RAs'... are increased thresholds or buffers warranted?



Backup Slides

TCAS-II Display Integration

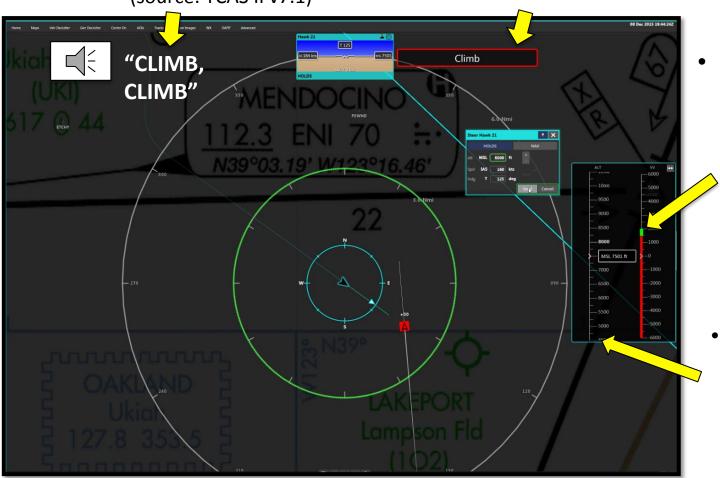


Auditory Alert

 RA sense presented aurally (source: TCAS II v7.1)

Text Based

 RA sense shown in text box next to Baseball Card



TCAS Vertical Rate Guidance

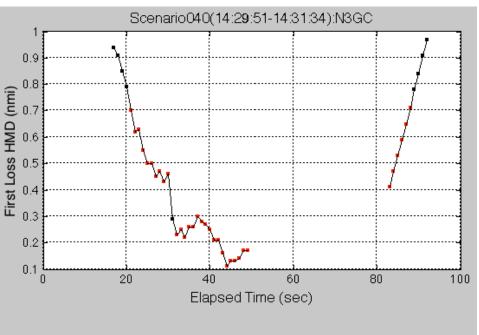
- Presented withinVVI
- Green = desiredvertical speed
- Red = vertical speed to avoid

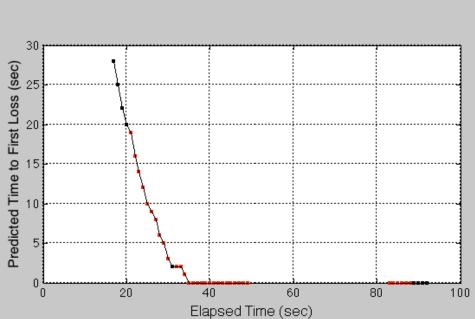
Inhibit DAA Altitude Guidance

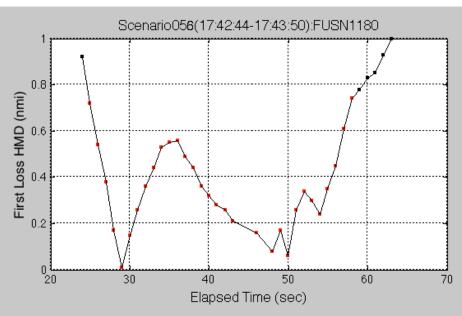


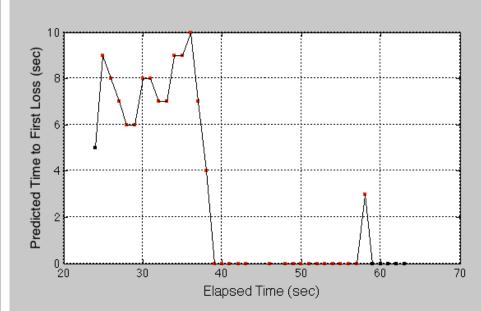
No Corrective Alert Prior to Warning Alert











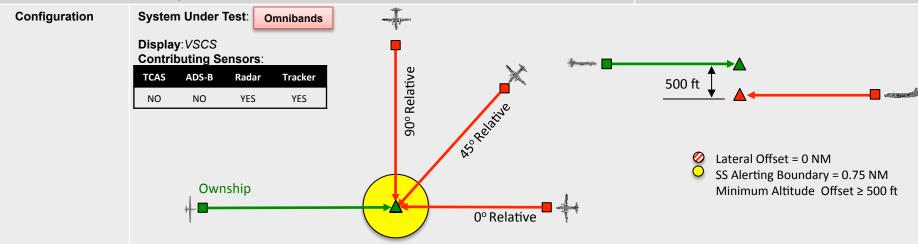




FT4 TEST OBJECTIVES - SSIWest.1.e (FT4 encounters: 013-015)

Objective 1: Validate DAA requirements in stressing cases that drive MOPS requirements: (e) low speed intruder (mitigated, <10kft MSL)

ACES Simulation Results – Common non-cooperative encounters



Test Objectives (TO)

- 1. Validate DAA alert timing allows pilot sufficient time to assess options and maneuver to remain well clear.
- 2. Validate DAA guidance to UAS pilot is appropriate (reasonable, timely & responsive).
- 3. Validate selected DAA maneuver results in remaining well clear, and removal of the alert once clear of threat.

Success Criteria (retest if criteria not met)

- 1. A timely (corrective initially) DAA alert is provided to the UAS pilot.
- 2. UAS pilot maneuvers in response to DAA alert.
- 3. Data collected: LVC log file, SAAProc log files, TCAS log files for ownship and intruder, SAAP file ('raw' surveillance data).

Test Method

- MANUEVER: Pilot to manually select and execute 'minimum' maneuver at edge of band.
- Aircraft speeds (non-accelerating): Ownship 120 KGS, Intruder 100 KGS
- Encounter Length: 2 min (IP to CPA)
- Stable Conditions: stable at the IP (2 minutes prior to CPA)
- Test Termination Criteria: targets diverging, range > 0.75 nmi., no DAA alerts displayed.
- Climb/Roll/Pitch Rates: 0/0/0
- Tolerance: ± 8 sec, ± 5 kts at IP crossing.

Evaluation Criteria (Post-test analysis to determine if test objectives are met)

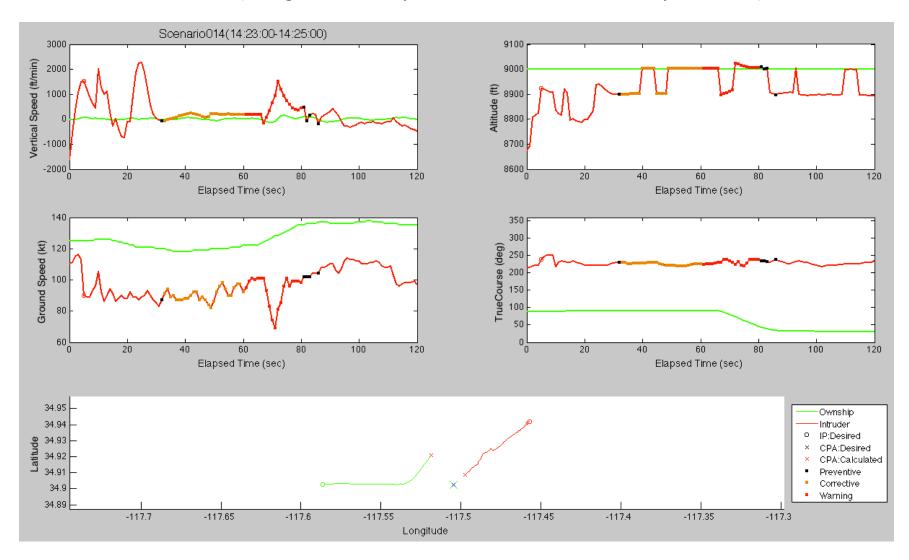
- 1. UAS pilot receives DAA corrective alert with associated guidance
- 2. UAS pilot maneuvers in response to DAA alert and remains well clear of intruder.
- 3. DAA alert(s) and guidance are removed once ownship is clear of threat.

Additional Information





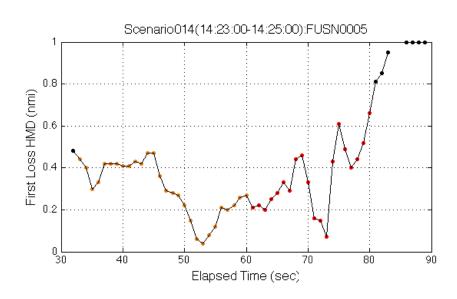
Scenario 014 (mitigated, low-speed, 45°, LoWC @ 78s elapsed time)

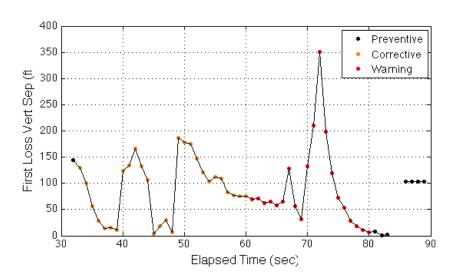


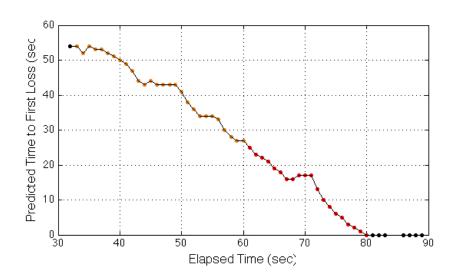


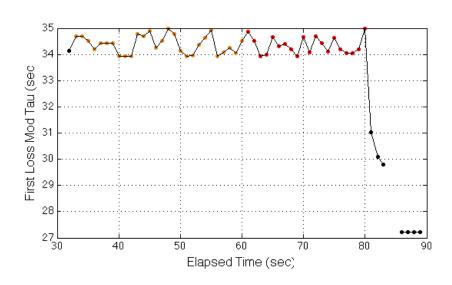


Scenario 014 (mitigated, low-speed, 45°, LoWC @ 78s elapsed time)





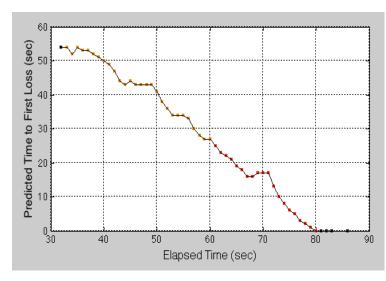




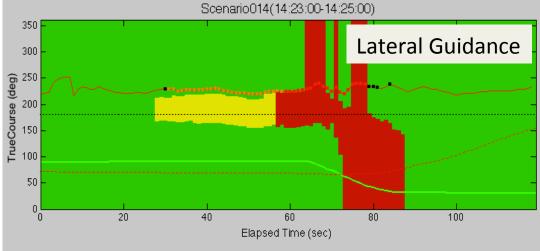


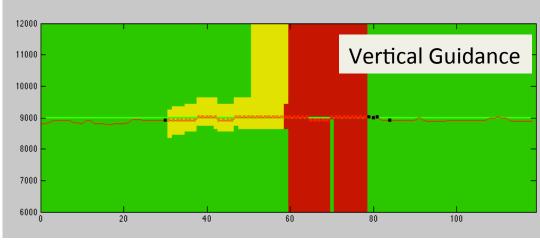


Scenario 014 (mitigated, low-speed, 45°, LoWC @ 78s elapsed time)



- Moderate ground speed and heading variance following ownship maneuver.
- LoWC for 2s @ >0.5nm HMD and with WCR guidance

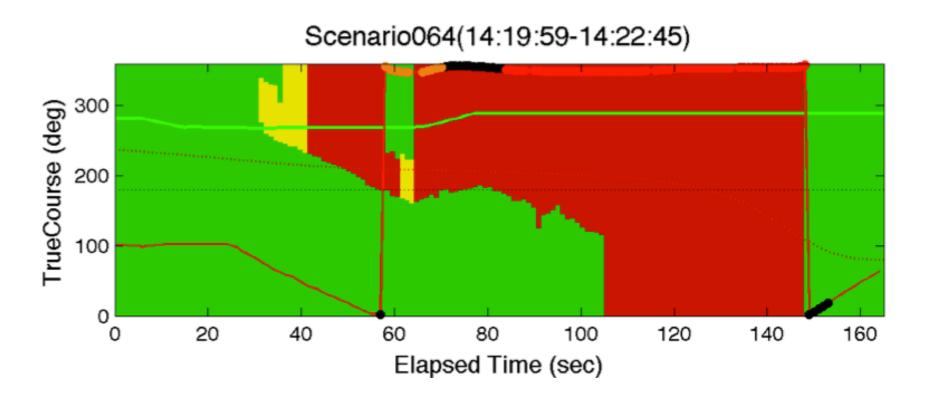






Mitigated LoWC: FT4 Procedural

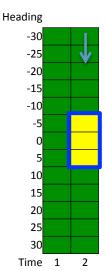


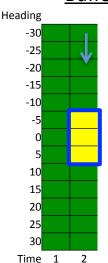




- In the presence of noisy sensors, the sizes and alert level within the bands may "jitter"
 - E.g. Yellows bands jumping from 30 degrees to 50 degrees back to 30 degrees in a matter of seconds
- Implemented buffering algorithm to keep the more conservative DAA guidance persistent over some user-specified time
- In example to the right, time window is 4
- Pro: "stable" guidance
- Con: over conservative in maneuver guidance, e.g. showing +/- 15 degree turn as necessary but +/- 5 degrees would be sufficient

Instantaneous guidance

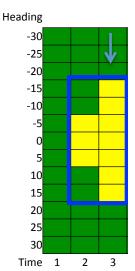


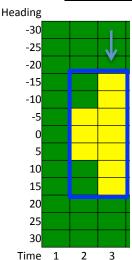




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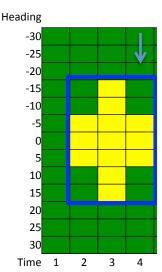


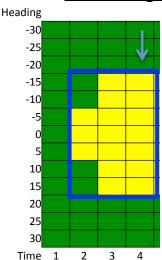




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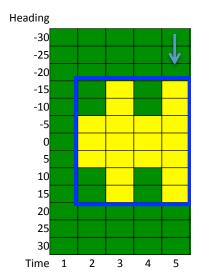


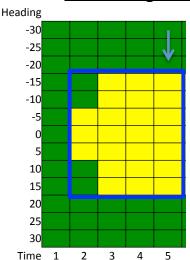




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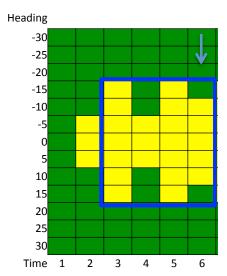


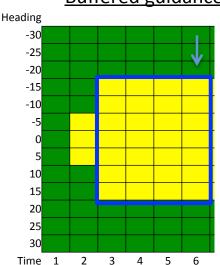




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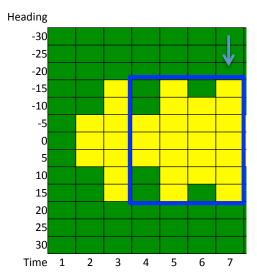


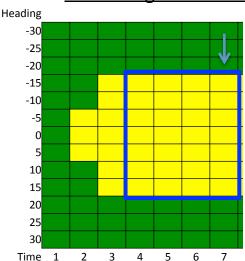
Head of queue is displayed to pilot



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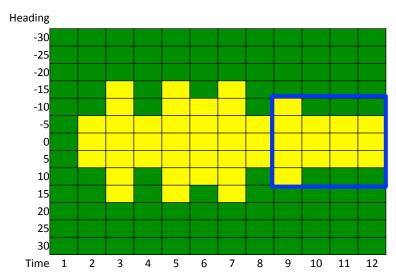


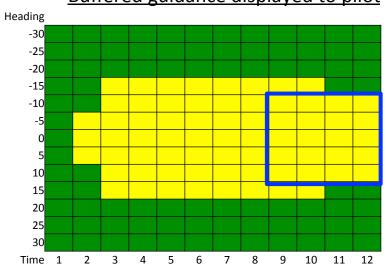
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